## Composite Technologies for Science Missions

NASA Marshall Space Flight Center
Engineering Directorate – Materials & Processes Laboratory
Non-Metallic Materials Division – Advanced Manufacturing Branch (EM42)
Chelsea Pickett



## Material Advantages

#### **Tailored materials properties**

Composites take advantage of different material properties to minimize weight while maximizing:

- Strength

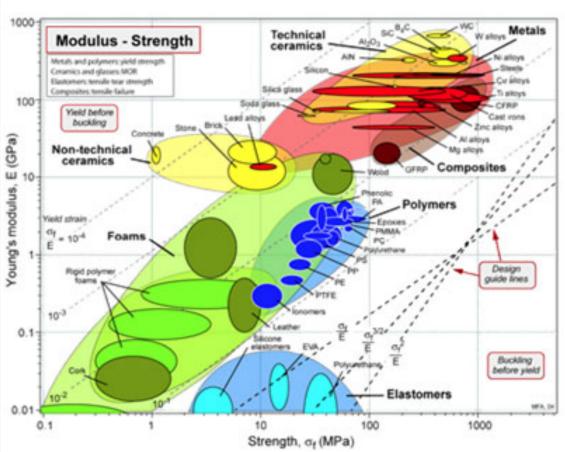
Primary (design drivers)

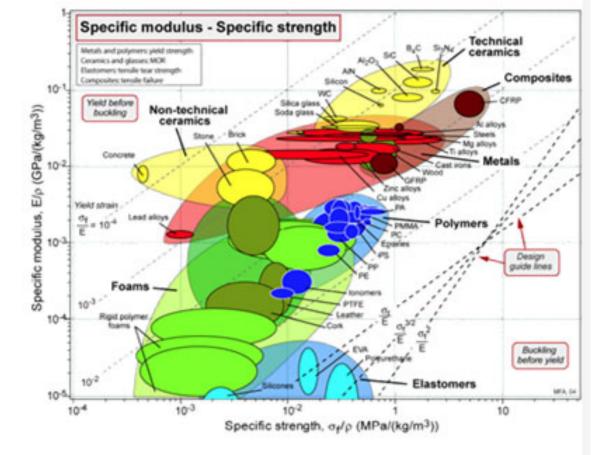
- Stiffness

- Radiation resistance > Secondary - can also be addressed

- Corrosion resistance

with coatings (个\$)







Graphite-based composites have zero to little coefficient of thermal expansion (CTE), allowing for simple designs in dynamic conditions



Structures can be designed to provide **protection from electrostatic discharge** (ESD)

## Design Considerations

Monolithic design: reduce part count

→ reduce cost

→ reduce risk of joint failure

#### **Brittle behavior:**

- + (pros) rigid structure with minimal creep, deflection
- (cons) low impact resistance, little warning of failure

**Natural Vibration**: tailor to specific frequency to mitigate communications issues as well as mechanical effects



**Conformation**: complex geometries are possible thanks to a large variety of textiles. No secondary/subtractive processing required.

→ Note: springback is a potential issue but can be mitigated with layup and tool design

#### Composite Textile Terminology

Layup: series of layers of 2D material Preform: single 3D textile product Weaves: 2D wrapping of fibers Knits: 3D wrapping of fibers

Tow: bundle of fibers
Tape: flattened tows
Filament: single fiber
or synonymous
with tow

# Science Mission Applications



**Vehicles**: rovers, landers, flying probes

Shells and platforms can be made using automated and/or manual techniques. Sandwich composites can be designed to provide additional stiffness.

Struts and other cylindrical bodies can be made using filament-winding.

Tanks: fuel, hydraulics



In lieu of traditional metallic tanks, all-composite or lined composite tanks can be made using the same technique as struts: filament-winding. Composite-overwrapped pressure vessels (COPVs) are common in modern designs.

Non-metallic fuel tank and support structure can be designed for thermal isolation.

## Manufacturing

Large-spanning structures, such as vehicle bodies, fairing and battery covers using **Automated Fiber Placement**. Automation reduces cost and variation.



Hand layup facilitates use of unique designs, textiles and consolidation/cure methodology.

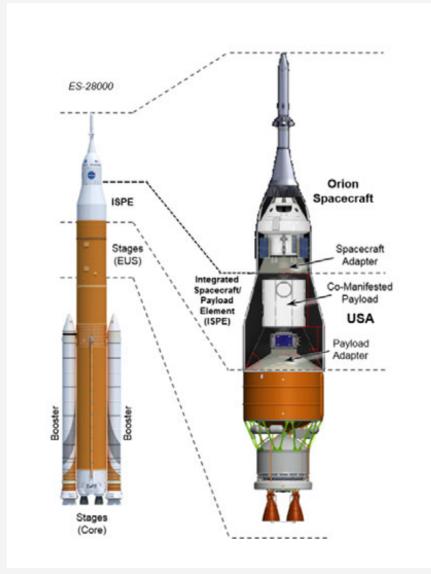
Cylindrical structures with high pressure requirements can be made using **filament winding**, which applies tension to the fibers during placement. Fibers are laid in helical and hoop patterns to distribute triaxial stresses.

Components such as **liners** and **nozzles** can be integrated for structural and/or chemical stability.

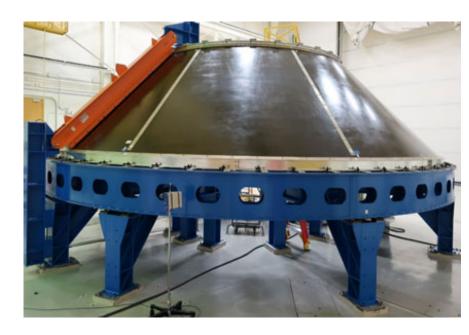




### Examples from NASA MSFC

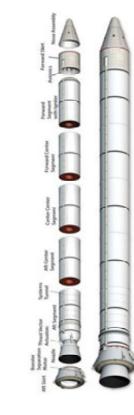


SLS Block 1B Payload Adapter sandwich panels assembled using bonded laminates



SLS BOLE Segment
Unlined, filament-wound case





CTE Bonded Joints
Hand-laid pi joint
and preform
C-channel

Composite
sandwich panel

C-channel

Pi joint