

David E. Smith, Principal Investigator

Maria T. Zuber, Deputy PI

LUNAR ORBITER LASER ALTIMETER

LOLA WILL DELIVER THE **MEASUREMENTS NEEDED TO RETURN ROVERS AND HUMANS** SAFELY TO THE MOON.

LOLA fully achieves three LRO measurement objectives and addresses two other. LOLA will provide all the data necessary to select intriguing, safe landing sites, while providing the reference system needed to navigate to those sites.

LOLA builds on extensive spaceflight heritage, including the Mercury Laser Altimeter (MLA) and the Mars Orbiter Laser Altimeter (MOLA). The LOLA measurement team has 15 years of altimetry experience that includes providing MOLA data to the Mars Exploration Rover site-selection teams.

LOLA will do at the moon what MOLA did at Mars, but with 3-5 times greater vertical accuracy and 32 times more frequent measurements along track.

MEASUREMENT OBJECTIVES

LRO Objectives	LOLA Measurement	LOLA Products		
• Global Geodetic Lunar Topography	Range	Global, regional, and local (met		
 Characterize Polar Region Illumination 	Range	scale) models of:		
 Image Permanently Shadowed Regions 	Range	Surface slopes Surface roughness		
• Assess meter-scale features to facilitate	Range, Roughness			
landing-site selection		Surface brightness		
 Identify near-surface water ice 	Reflectance	Improved lunar gravity model		

LOLA fully achieves bold objectives

OPERATIONS OVERVIEW

The LOLA instrument pulses a single laser through a Diffractive Optical Element (DOE) to produce five beams that illuminate the lunar surface. For each beam, LOLA measures time of flight (range), pulse spreading (surface roughness), and transmit/return energy (surface reflectance). With its 2-D spot pattern, LOLA unambiguously determines slopes along-track and across-track.

In a 50km polar orbit, pulsing the laser at 28 Hz creates an ~50m-wide swatch of five topographic profiles. Swaths will have 1.25km separation at the equator, with [complete polar coverage beyond +/-86 degrees latitude.] Raw measurements are transmitted to Earth for analysis.

LOLA's robust link margin provides ample reserve to accommodate uncertainties in lunar surface roughness and albedo, while providing operational flexibility to the LRO mission







Key Instrument Parameters						
Mass	9.6 kg					
Volume	0.45 m length					
	0.31 m width					
	0.35 m height					
Power	$26.2 \ W \ (\texttt{29.9 W turn-on peak})$					
Data Rate	10 kbps					
Link Margin	5.4dB at 50 km orbit					

LOLA Products				
Global, regional, and local (meter-				
scale) models of:				
 Geodetic Topography 				
Surface slopes				
 Surface roughness 				
Surface brightness				





- Experience and personnel carry over from Mercury Laser Altimeter (MLA)
- Management organization is simple with clear lines of authority.



E/PO OBJECTIVES

- Partner with existing dissemination networks, including those of the MESSENGER mission, to reach a broad and diverse audience
- Inspire the next generation through the infusion of lunar and planetary data into the classroom (K-12 and undergraduate)
- Include lunar 3-D topographic models in education kits targeting a variety of audiences, including special-needs students, in a variety of venues

SCHEDULE & COST						
LOLA	2004	2005	2006	2007		35cm
Project Milestones		♦ ♦ SRR PDR	¢ CDR	PER PSR Del		
Design & Development	-					
I&T						
Contingency						45 cm - 31 CM
Spacecraft I&T						LOLA MLA
Cost: (Contingency:	TOTA	AL:	· ·		The LOLA instrument is TRL 7 based on MLA heritage

SUMMARY High Evaluation Merit

Technically Feasible

- Strong heritage
- No New technology

Low Implementation Risk

• Experienced measurement and implementation teams

Fully addresses 3 LRO Measurement Objectives
Contributes 5 of the 7 LRO measurement datasets

• Backed by GSFC's exceptional personnel and facilities